

## HALT to Qualify Electronic Packages – A Proof of Concept\*

## Rajeshuni Ramesham Ph.D.

Jet Propulsion Laboratory Reliability Engineering, Office of Safety and Mission Success California Institute of Technology Pasadena, CA 91109

Tel.: 818 354 7190

E-mail: Roteshuni ramesham@jpl.nasa.gov

June 18, 2014

\*Copyright 2014 California Institute of Technology. Government sponsorship acknowledged.



# Acknowledgements

- The research work described in this paper was carried out at the Jet Propulsion Laboratory, Caltech, Pasadena, CA.
- NEPP program is highly appreciated for supporting me to work on HALT task. MSL Project has generated the interest in me in obtaining the experimental data for SMT packages in a short duration for present and future JPL/NASA projects.
- This paper was presented at the SPIE Reliability conference in Feb 2014.
- Thanks are due to
  - Dr. Chuck Barnes, Dr. Doug Sheldon, Mr. Mike Sampson, and Mr. Ken Label for their interest and support under NEPP packaging program.
  - Mr. Russel Kido of Practical Components for his support during the test board manufacturing.
  - Emilio Vazquez for his help in the data logging system.
  - John Forgrave and Mark Boyles for their comments/suggestions on the presentation.





- Life Testing Requirements
- Introduction
- Objectives
- Experimental Approach and Details
- Test Results
- Summary
- Acknowledgements



The following requirement is from the JPL Design Principles.

## Thermal cycling -

 Electronic hardware design shall be capable of surviving power ON-OFF temperature cycling and/or solar exposure cycling of <u>three times</u> the number of worst-case expected mission cycles with worst-case flight temperature excursions.



Mars

Rovers

Exploration

### Mars Family of Rovers and Lander: Pathfinder, Spirit/Opportunity, Curiosity, & Phoenix

Mars

Science

90 sol mission Test: 270 thermal cycles 28 days of testing

> 670 sol mission Test: 2010 thermal cycles 209 days of testing

2.000 pounds

7 sol mission Test: 21 thermal cycles 2.2 days of testing

Sojourner

Rover

90 sol missionTest: 270 thermal cycles28 days of testing

June 16, 2014



- **Introduction**
- Explore *a proof of concept* of Highly Accelerated Life Testing (HALT) technique to assess packaging designs for long duration deep space missions in a wide temperature range (-150°C to +125°C).
- HALT is a custom hybrid package suite of testing technique, such as extreme temperature down to -150°C to +125°C and vibration step processing up to 50g acceleration.
- HALT testing implements repetitive multiple-axis vibration combined with thermal cycles testing of the test article to precipitate defects.



Objective

- Predict reliability, failure mechanism and survivability of selected advanced electronic interconnect packages (Ball Grid Arrays, Flipchips, surface mount packages, etc.) for long duration deep space missions in a shorter duration.
- Reduce the development cycle time for improvements to the packaging design.



## **Experimental Approach and Details**

- Design and build advanced test boards that are considered useful for a variety of NASA projects.
- Select stress stimuli thermal cycling and/or all-axis vibration or both simultaneously
- Perform stress testing
- Analyze the test results and compare with the thermal cycling test data.
- Estimate the acceleration factor for package failures observed.





HALT Test Chamber (Shock and Thermal capability)

# Schematic diagram for the solder joint continuity monitoring HALT test system



### **Two Test Boards Were Tested**

Board #	Board Label : Test Point	Channel #	
1	BGA256 Left Top ± T1	1001	
1	BGA256 Left Top ± T2	1002	
1	BGA256 Left Bottom ± T1	1003	
1	BGA256 Left Bottom ± T2	1004	
1	PBGA388 Top ± A	1005	
1	PBGA388 Top ± B	1006	
1	PBGA388 Top ± C	1007	
1	PBGA388 Bottom ± A	1008	
1	PBGA388 Bottom ± B	1009	
1	PBGA388 Bottom ± C	1010	
1	CVBGA97 Top ± T	1011	
1	CVBGA97 Bottom ± T	1012	
1	QFP100 Left ± T	1013	
1	QFP100 Right ± T	1014	
1	QFP256 Left ± T	1015	
1	QFP256 Right ± T	1016	
1	QFP208 Left ± T	1017	
1	QFP208 Right ± T	1018	
1	MLF68 Top ± T	1019	
1	MLF68 Bottom ± T	1020	
2	BGA256 Left Top ± T1	1021	
2	BGA256 Left Top ± T2	1022	
2	BGA256 Left Bottom ± T1	1023	
2	BGA256 Left Bottom ± T2	1024	
2	PBGA388 Top ± A	1025	
2	PBGA388 Top ± B	1026	
2	PBGA388 Top ± C	1027	
2	PBGA388 Bottom ± A	1028	
2	PBGA388 Bottom ± B	1029	
2	PBGA388 Bottom ± C	1030	
2	CVBGA97 Top ± T	1031	
2	CVBGA97 Bottom ± T	1032	
2	QFP100 Left ± T	1033	
2	QFP100 Right ± T	1034	
2	QFP256 Left ± T	1035	
2	QFP256 Right ± T	1036	
2	QFP208 Left ± T	1037	
2	QFP208 Right ± T	1038	
2	MLF68 Top ± T	1039	
2	MLF68 Bottom ± T	1040	

Each test board has similar components. Some of the components were independently daisy chained.

BGA: Ball grid array PBGA: Plastic ball grid array CVBGA: Very thin chip array ball grid array QFP: Quad flat pack

**MLF: Micro Lead Frame** 

Source: Practical components data sheet



### **Test Boards in the HALT System**





### Two Test boards in the HALT chamber





#### **Experimental Details**



Temperature (125°C), g level (~10g, 30,<br/>40g) vs. test duration of 1 hourTemperature (-150°C), g level (~40g) vs. test<br/>duration of 1 hour



#### Test Results summary: Various test runs with variations of temperature, acceleration, and test duration

S/N	Temperature, C	Acceleration, g	Test Duration	Observation
1	+25oC to -125oC	5g		No anomalous behavior observed during this test.
2	@50C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
3	@75C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
4	@100C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
5	@125C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
6	@0C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
7	@-25C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
8	@-50C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
9	@-75C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
10	@-100C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
11	@125C	10, 15, 20, 30, 40, and 50g	10 minutes at each g level	No anomalous behavior observed during this test.
12	@25C	10, 15, 20, 30, 40, and 50g	60 minutes at each g level	No anomalous behavior observed during this test.
13	@75C	10, 15, 20, 30, 40, and 50g	60 minutes at each g level	No anomalous behavior observed during this test.
14	@100C	10, 15, 20, 30, 40, and 50g	60 minutes at each g level	No anomalous behavior observed during this test.
15	@125C	10, 15, 20, 30, 40, and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance
16	@25C	10, 30 and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance
17	@75C	10, 30 and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance
18	@125C	10, 15, 20, 30 and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance
19	@20C	10, 20, 30 and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance
20	@-50C	10, 20, 30 and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance
21	@-100C	10, 20, 30 and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance
22	@-150C	10, 20, 30 and 50g	60 minutes at each g level	Failures were observed during this test run. Observed change in the daisy-chain resistance



Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Time –

No anomalies were observed in all the daisy-chains of advanced electronic packages since the resistance is constant vs test time



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



#### Time —

Resistance of various daisy chains vs., time at a given dynamic load of g particularly the daisy-chain resistance anomalous behavior of PBGA Package



#### Test Results: Resistance of PBGA daisy chain vs. time at a given dynamic load of g and temperature



Resistance of PBGA daisy chain vs., time at a given dynamic load of g particularly (corresponding to channel #1009). Daisy chain resistance change was highly intermittent due to the failure of a PBGA388 package (on board#1).



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Resistance of PBGA daisy chains vs., time at a given dynamic

load of g particularly (corresponding to channel #1009 (board 1), 1025, 1024, 1027 (board#2)). Daisy chain resistance change was highly intermittent due to the failure of a PBGA388 package on board#1 and small variations in daisy chain resistance of BGA256, PBGA388 of board#2.



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Time

PBGA388 (channel 1009) has failed completely and daisy chain resistance is infinity or open circuit.

Ram



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



PBGA388 (channel 1009) has failed completely and daisy chain resistance is very high. Resistance is normal when no stress is applied on test board.



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



#### **Resistance of some of the daisy chains were responding**



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Change is daisy chain resistance of QFP208, QFP100, QFP100, MLF68 as a result of combined stresses of thermal and dynamic loads on board #1.



Resistance, Ohms -

National Aeronautics and Space Administration Jet Propulsion Laboratory California Institute of Technology

Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Daisy chain resistance of QFP256 on board#2 and QFP256 on board#1. Resistance fluctuations of QFP on board #2 are more over the similar QFP256 on board#1.

June 16, 2014



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Daisy chain resistance of MLF68 on board 1 and board 2. Slight fluctuations or intermittent changes in a daisy chain resistance of MLF68 on board #2. Similarly PBGA388 on board#2.



Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Captured a screen shot of daisy chain resistance of PBGA on board #1 channel 1009 during the test. The curve indicates the intermittent failure of daisy chain resistance of PBGA388



#### Test Results: All the test data was normalized to worst case Delta T temperature and shock level.

	Temperature, C	T2, min	T2, hours	G1, g	G2, g	T2, hours	
	50	10	0.16667	10	50	5.60326E-06	
	50	10	0.16667	15	50	7.50628E-05	
	50	10	0.16667	20	50	0.000473187	
	50	10	0.16667	30	50	0.006338942	
	50	10	0.16667	40	50	0.020050025	
	50	10	0.16667	40	50	0.039939923	
	50	10	0.16667	50	50	0.166666667	
	75	10	0.16667	10	50	5.60326E-06	
	75	10	0.16667	5	50	7.50628E-05	
	75	10	0.16667	20	50	0.000473187	
	75	10	0.16667	30	50	0.006338942	
	75	10	0.16667	40	50	0.039959925	
	75	10	0.16667	40	50	0.039939923	
	/5	10	0.16667	50	50	0.166666667	
	100	10	0.16667	10	50	5.60326E-06	
	100	10	0.16667	15	50	7.50628E-05	
	100	10	0.16667	20	50	0.000473187	
	100	10	0.16667	30	50	0.006338942	
	100	10	0.16667	40	50	0.039959925	
	100	10	0.16667	50	50	0.166666667	
	125	10	0.16667	10	50	5.60326E.06	
	123	10	0.16667	10	50	3.60326E-06	
	125	10	0.16667	15	50	7.50628E-05	
	125	10	0.16667	20	50	0.000473187	
	125	10	0.16667	30	50	0.006338942	
	125	10	0.16667	40	50	0.039959925	
	125	10	0.16667	50	50	0.166666667	
	0	10	0.16667	10	50	5.60326E.06	
	<u> </u>	10	0.16667	15	50	7 50628E 05	
	5	10	0.10007	15	50	7.30028E-05	
	0	10	0.16667	20	50	0.000473187	
	0	10	0.16667	30	50	0.006338942	
	0	10	0.16667	40	50	0.039959925	
	0	10	0.16667	50	50	0.166666667	
	-25	10	0.16667	10	50	5.60326E-06	
	-25	10	0.16667	15	50	7.50628E-05	
	-25	10	0.16667	20	50	0.000473187	
	-23	10	0.16667	20	50	0.000473187	
	-25	10	0.16667	30	50	0.006338942	
	-25	10	0.16667	40	50	0.039959925	
	-25	10	0.16667	50	50	0.166666667	
	-50	10	0.16667	10	50	5.60326E-06	
	- 50	10	0.16667	15	50	7.50628E-05	
	-50	10	0.16667	20	50	0.000473187	
	-50	10	0.16667	20	50	0.006338042	
	-30	10	0.16667	30	50	0.000338942	
	-50	10	0.16667	40	50	0.039959925	
	-50	10	0.16667	50	50	0.166666667	
	-75	10	0.16667	10	50	5.60326E-06	
	-75	10	0.16667	15	50	7.50628E-05	
	-75	10	0.16667	2.0	50	0.000473187	
	-75	10	0.16667	30	50	0.006338942	
	75	10	0.16667	40	50	0.020050025	
	-73	10	0.16667	40	50	0.039939923	
	- 75	10	0.16667	50	50	0.166666667	
	-100	10	0.16667	10	50	5.60326E-06	
	-100	10	0.16667	15	50	7.50628E-05	
	-100	10	0.16667	20	50	0.000473187	
	- 100	10	0.16667	30	50	0.006338942	
	-100	10	0.16667	40	50	0.039959925	
	100	10	0.16667	50	50	0.166666667	
	-100	10	0.16667	30	50	0.100000007	
	-125	10	0.16667	10	50	5.60326E-06	
	-125	10	0.16667	15	50	7.50628E-05	
	-125	10	0.16667	20	50	0.000473187	
	-125	10	0.16667	30	50	0.006338942	
	-125	10	0.16667	40	50	0.039959925	
	-125	10	0.16667	50	50	0.166666667	
	25	60	1	50	50	1	
	23	60	-	10	50	2 261657 25	
	75	60	1	10	50	3.36196E-05	
	75	60	1	30	50	0.03803365	
	75	60	1	50	50	1	
	125	60	1	10	50	3.36196E-05	
	125	60	1	30	50	0.03803365	
	125	60	1	50	50	1	
	25	60	1	10	50	3.36196E-05	
	25	60	-	30	50	0.03802265	
	23	60	-	50	50	0.03803303	
	20 7.5	60	1	30	50	2 261657 25	
	75	60	1	10	50	3.36196E-05	
	75	60	1	30	50	0.03803365	
	75	60	1	50	50	1	
	125	60	1	10	50	3.36196E-05	
1	125	60	1	30	50	0.03803365	
1	125	60	1	50	50	1	
	20	60	1	10	50	1 2 26106E 05	
	20	60	1	10	50	3.36196E-05	
	20	60	1	30	50	0.03803365	
1	20	60	1	50	50	1	
1	-50	60	1	10	50	3.36196E-05	
1	-50	60	1	30	50	0.03803365	
	-50	60	1	50	50	1	
	-100	60	1	10	50	3.36196E-05	
1	100	60	1	30	50	0.03802265	
	-100	60	1	50	50	0.03803365	
1	- 100	60	1	50	50	1	26
1	-150	60	1	10	50	3.36196E-05	20
	-150	60	1	30	50	0.03803365	
	-150	60	1	50	50	1	

5<sup>th</sup> NEP<del>P Electronic Technology Workshop</del>



## **Extreme Temperature Thermal Cycling Profile**



Time, hours



	First Failure
Electronic Package	Thermal Cycle
Туре	Number
PBGA388	711
MLF68	235
QFP100	711
QFP208	324
BGA256	235

Failure of advanced electronic packages vs. number of thermal cycles from -185°C to +125°C



Summary-1

- Studied a *proof of concept* of HALT technique to assess fatigue reliability of electronic packaging designs:
  - ✓ temperature range:  $-150^{\circ}$ C to  $+125^{\circ}$ C
  - $\checkmark\,$  acceleration range of up to 50g.
- The test boards were subjected to various g levels, dwell durations, and the hot and cold temperature levels. Advanced electronic packages have shown signs of continuity problems.
  - ✓ plastic ball grid array (PBGA)
  - ✓ ball grid array (BGA)
  - ✓ micro-lead-frame (MLF)
  - ✓ quad flat-pack (QFP)
- In this very preliminary study, PBGA package was completely open where as other package designs have shown signs of continuity variations.





- The failure of the PBGA occurred within 12 hours of the start of accelerated testing using dynamic and thermal loads.
- The PBGA package failed during the independent thermal cycling test:
  - ✓  $-185^{\circ}$ C to  $+125^{\circ}$ C after 711 thermal cycles (test data)
    - 959 cycles of -150°C to +125°C, extrapolated from the test data where m=2.5)
  - ✓ Each thermal cycle requires 2.33 hours and a total test time to fail PBGA was 2,237 hours (or ~3.1 months) due to thermal cycling fatigue alone.
- HALT technique required only 12 hours to fail
  - ✓ indicates that there is an acceleration factor of ~186 times (more than 2 orders of magnitude) to fail the same PBGA component.
- ✓ We do not know the failure mechanism of PBGA failures yet in HALT technique and is yet to be determined. Relationship between failure mechanism and HALT test parameters will be addressed in the future efforts. DPA and X-ray imaging techniques and others will be employed to understand the failure mechanisms.
- One can determine in this preliminary study that the life of the PBGA component in less than 12 hours of testing using HALT instead of testing for 3.1 months in thermal cycling alone.



# Acknowledgements

- The research work described in this paper was carried out at the Jet Propulsion Laboratory, Caltech, Pasadena, CA.
- NEPP program is highly appreciated for supporting me to work on HALT task. MSL Project has generated the interest in me in obtaining the experimental data for SMT packages in a short duration for present and future JPL/NASA projects.
- This paper was presented at the SPIE Reliability conference in Feb 2014.
- Thanks are due to
  - Dr. Chuck Barnes, Dr. Doug Sheldon, Mr. Mike Sampson, and Mr. Ken Label for their interest and support under NEPP packaging program.
  - Mr. Russel Kido of Practical Components for his support during the test board manufacturing.
  - Emilio Vazquez for his help in the data logging system.
  - John Forgrave and Mark Boyles for their comments/suggestions on the presentation.

## **Back-up slides**



## Why do we need? NASA

- ✓ Titan (-180°C, for a proposed Titan *in-situ* mission)
- ✓ Europa (-160°C, for a proposed Europa surface and subsurface mission)
- ✓ Asteroids (-185°C, MUSES-CN project)
- ✓ Comets (-140°C, for a proposed comet nucleus sample return)
- ✓ Earth's moon (recorded temperature on the moon: -233°C to +123°C, moon mineralogy and mapper, M<sup>3</sup>)
- ✓ Mars Exploration Rover, MER (-120°C to +85°C)
- ✓ MSL (-135°C to +85°C) large diurnal temperature change/swing from day to night.
- ✓ Planetary protection requires the hardware to be baked at +125°C for 72hrs to kill microorganisms to avoid any biological contamination, especially for sample return missions.
- ✓ NASA standard thermal cycling temperature range varies from -55°C to +100°C for 200 thermal cycles
- ✓ HALT system capability is -160°C to +200°C
- ✓ The present HALT package reliability research study has encompassed the temperature range of -150°C to +125°C which cover only some potential future NASA space missions.



#### Test Results: Resistance of various daisy chains vs. time at a given dynamic load of g and temperature



Resistance of various daisy chains vs., time at a given dynamic load of g particularly the daisy-chain resistance anomalous behavior of BPGA Package



#### **Experimental Details-3**



Temperature (25°C and 75°C) and g level (10g, 30g, an 40g) vs. test duration of 0.5 hour Temperature (75°C, 100°C, 125 °C) and g level (10g, 30g, an 40g) vs. test duration of 10 minutes



#### **Experimental Details-1**



Temperature (125°C), g level (~10g, 30, 40g) vs. test duration of 1 hour

Temperature (0°C, 50°C), g level (~10g, 30, 40g) vs. test duration of 1 hour



#### **Experimental Details-2**





Temperature (-150°C), g level (~10g) vs. test duration of 1 hour Temperature (-150°C), g level (~40g) vs. test duration of 1 hour



Estimating dynamic/thermal load fatigue

 $N_1 (G_1)^{b-dynamic} = N_2 (G_2)^{b-dynamic}$ 

 $N_1 (\Delta T_1)^{b-\text{thermal}} = N_2 (\Delta T_2)^{b-\text{thermal}}$ 

Estimating dynamic and thermal load fatigue (assume the stresses are linear in nature)

$$N_1 \left[ (G_1)^{b\text{-dynamic}} + (\Delta T_1)^{b\text{-thermal}} \right] = N_2 \left[ (G_2)^{b\text{-dynamic}} + (\Delta T_2)^{b\text{-thermal}} \right]$$

 $T_1 (G_1)^b = T_2 (G_2)^b$ 

T: Test Time, N: Number of stress cycles, b: fatigue exponent (thermal: 2.5; vibration: 4) for 63/37 Pb/Sn solder