



***Status of: 'Guidelines for  
Space Qualification of GaN  
HEMT Technology'***

***John Scarpulla  
Caroline Gee***

***17 June, 2020  
NASA Electronics Parts & Packaging  
Program (virtual conference)***



# Background

## *Status of space qual of GaN RF/Microwave HEMT technology*

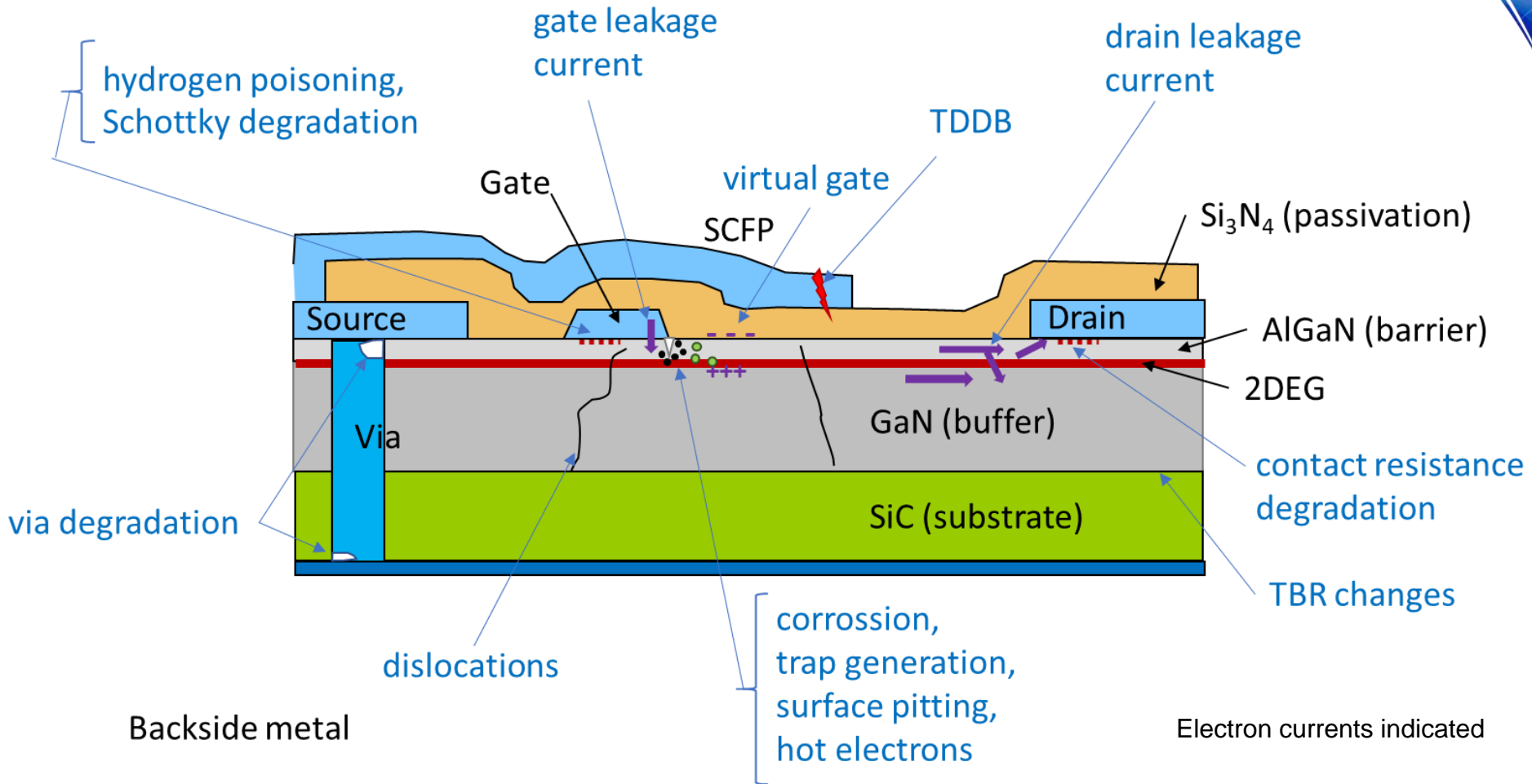
- GaN RF/microwave HEMTs and MMICs are beginning to find their way into Class A and B space missions
  - *GaN HEMT advantages*
    - High power density -- Watts/mm of gate length
    - High voltage capability – eases RF matching
    - Radiation hard -- >10Mrad
  - *GaN HEMT issues*
    - Trapping phenomena causes new problems
    - Current collapse, pulse-to-pulse instability, low frequency noise etc.
    - Piezoelectric effects at high voltage may generate even more traps
- No official GaN government standards presently exist for qualification in space
  - *Existing standards cover Si, GaAs devices*
- Aerospace TOR (Tech. Operating Report) is an attempt to fill this gap
  - *Weekly working group meetings held from 8/19 – 12/19*
  - *100 participants from GaN fabs, aerospace firms, industry, government*
  - *Previous draft document received numerous comments*

***Please request TOR TOR-2018-00691 “Guidelines for Space Qualification of GaN HEMT Technologies” J. Scarpulla, C. Gee***



# Failure Mechanisms in GaN HEMTs

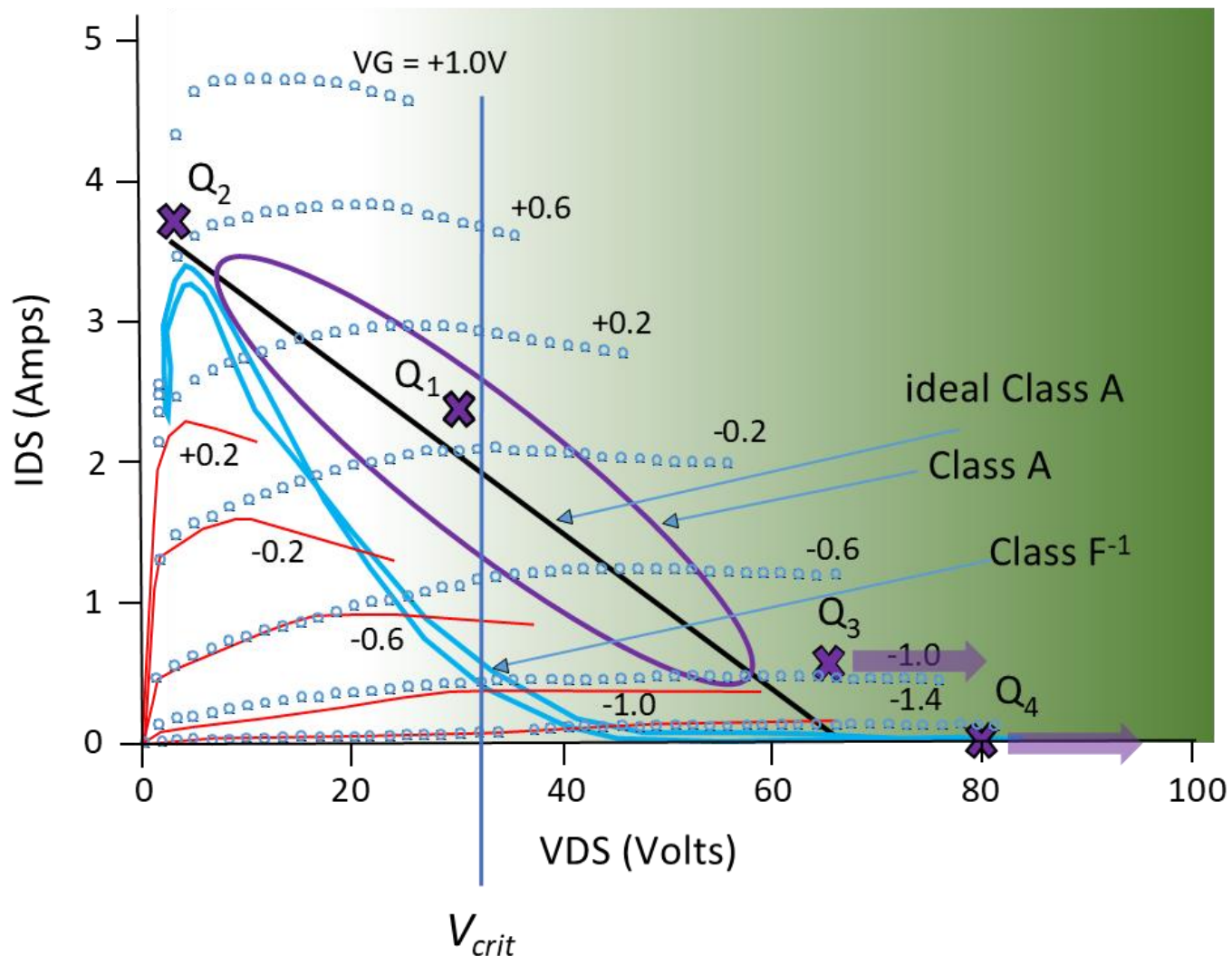
Understanding of the failure modes guides qual testing



**Different failure modes operate in different electrical regimes**

# The IV Plane

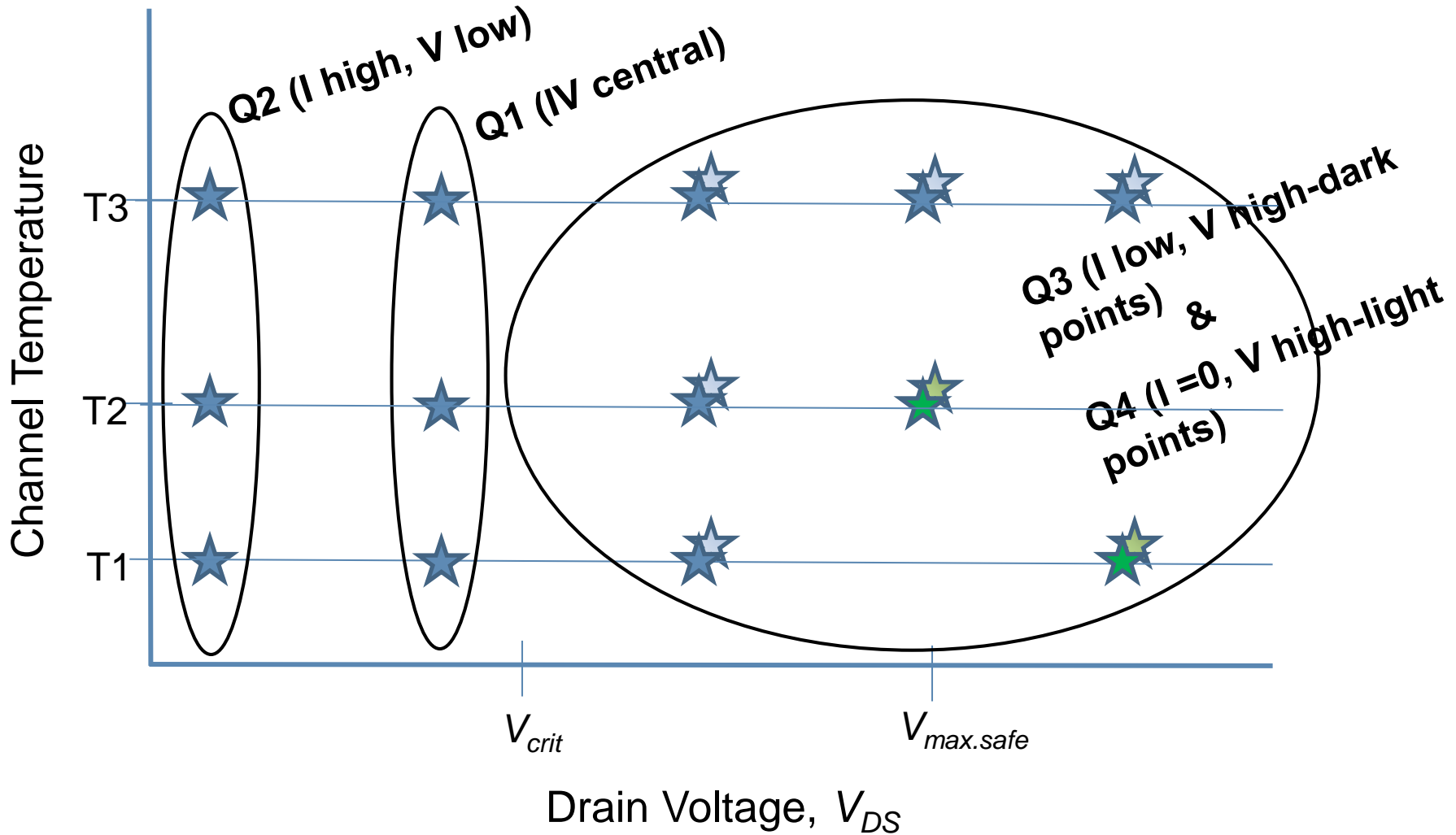
Recommendations for accelerated life testing (ALT)



**The “DC vs. RF” question – can DC rel testing be sufficient for an RF device?**

# DC Reliability Testing Scheme

including temperature and voltage accelerants (2-dimensional problem)

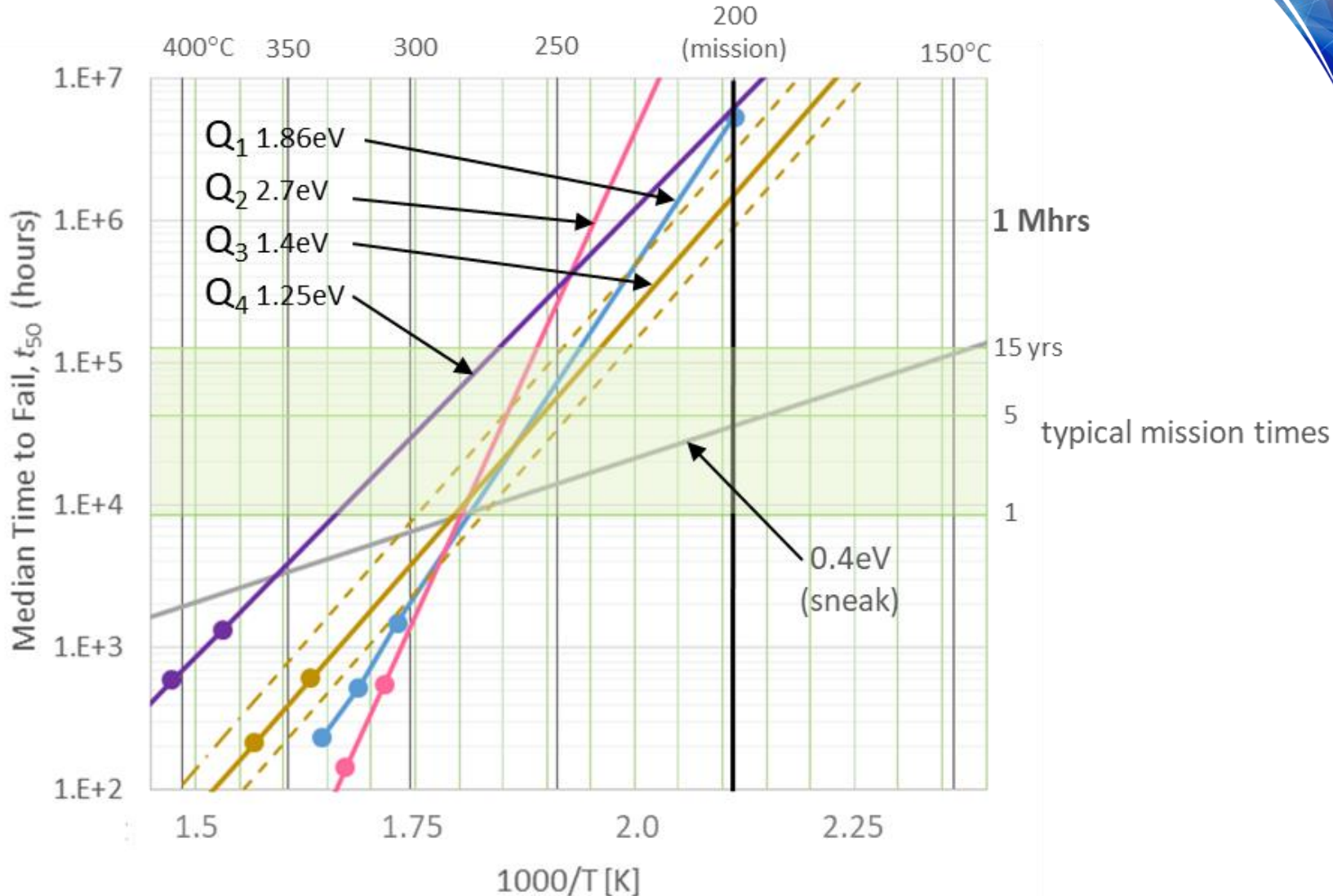


**Optimum test plan includes all four “Q-points”**



# Multiple Q-points at different temperatures

Activation energies differ

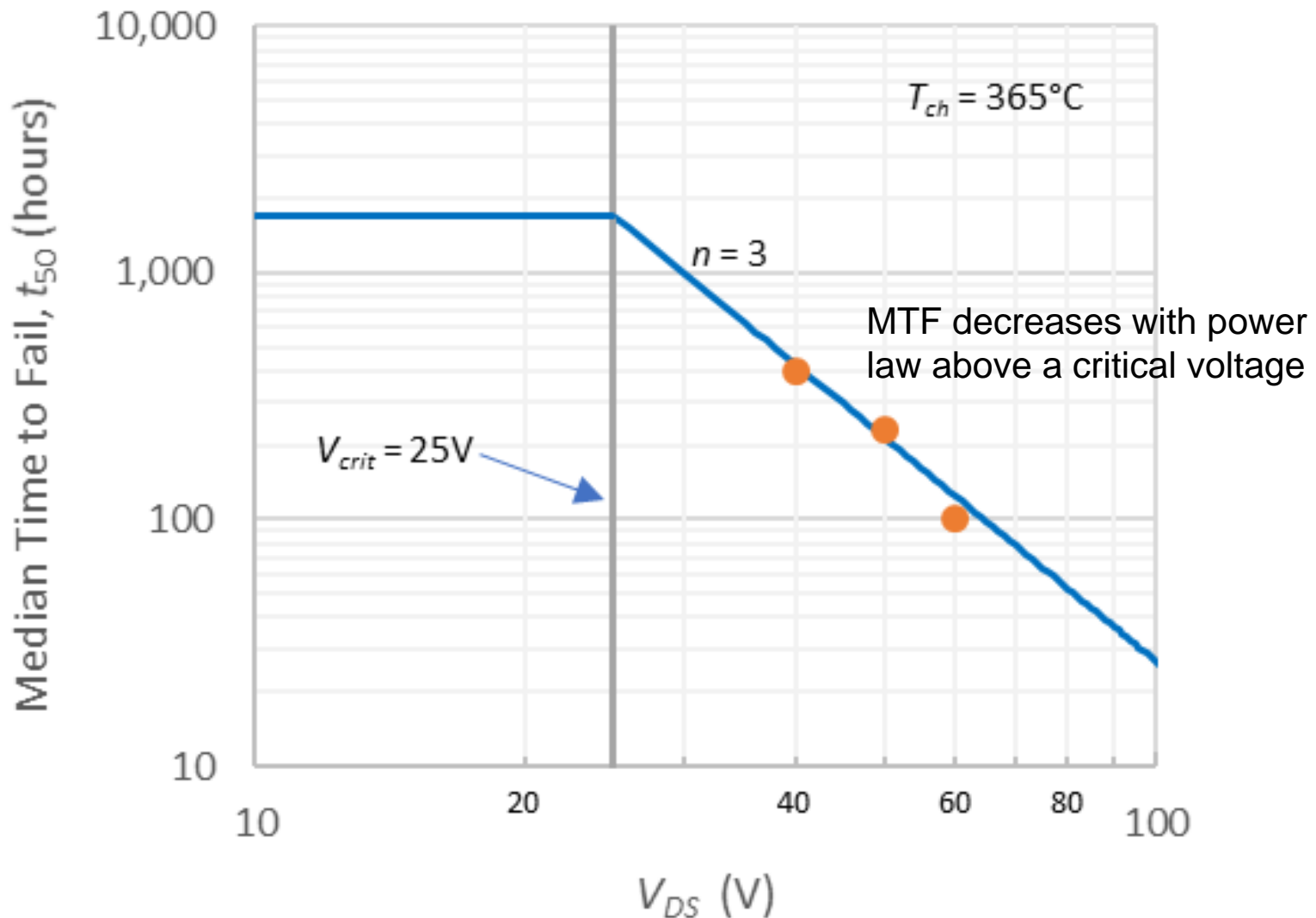


**Arrhenius plots emphasize temperature acceleration .... What about voltage acceleration?**



## Q4 (off state) Voltage accelerated test

(at constant accelerated temperature)



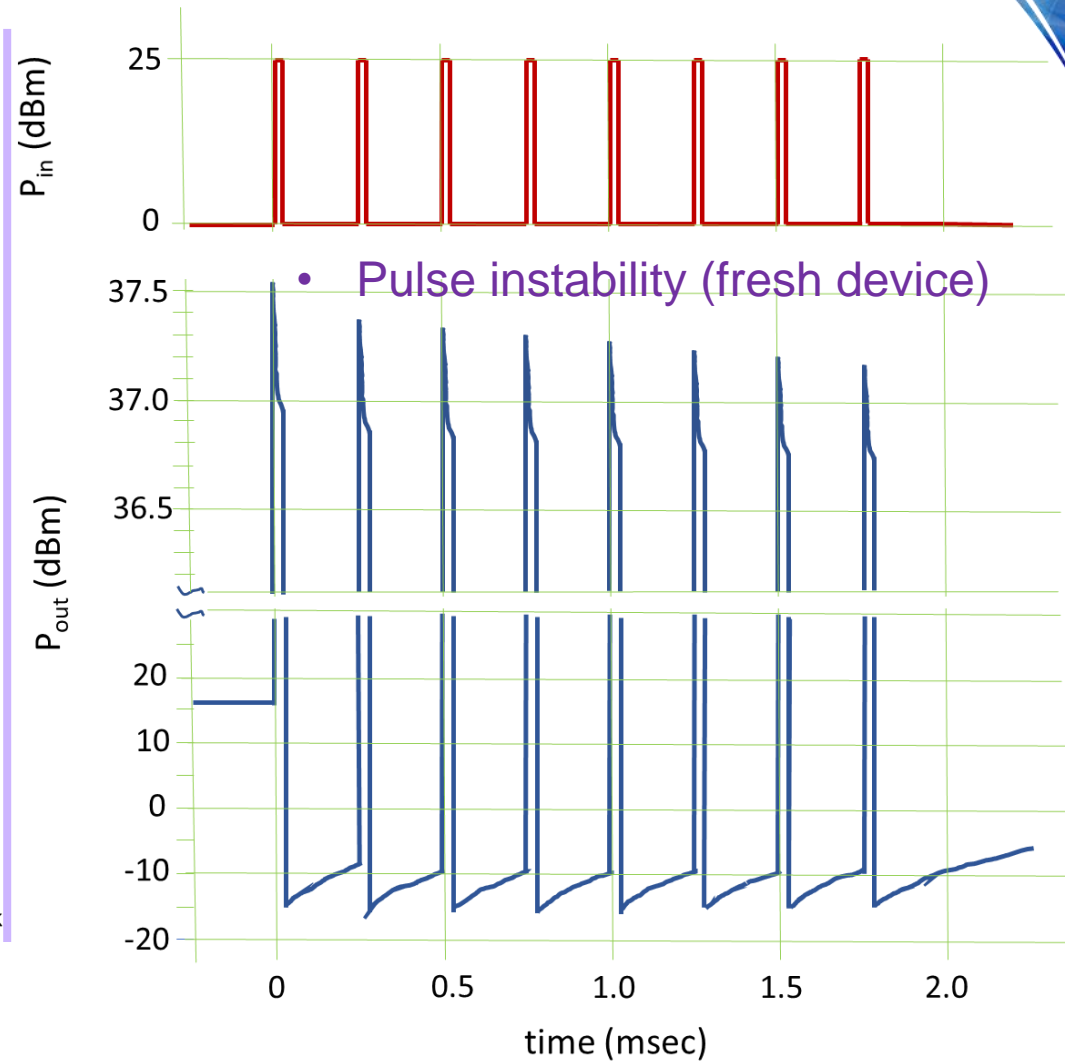
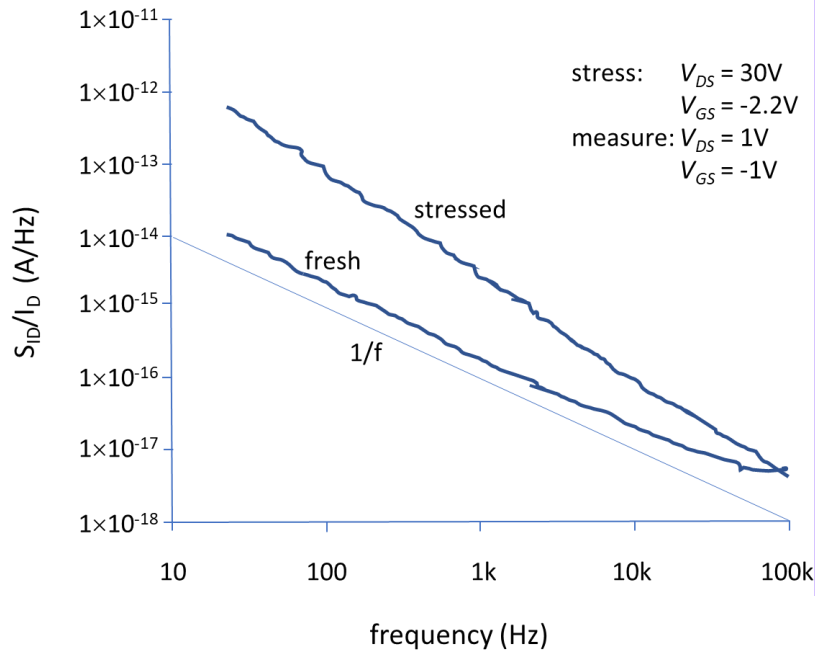
**Not all HEMTs exhibit a voltage dependence of accelerated failure time**



# Why traps are important

Trapping causes noise, pulse distortion and instability

- Low frequency noise increase after 1,000 hr stress



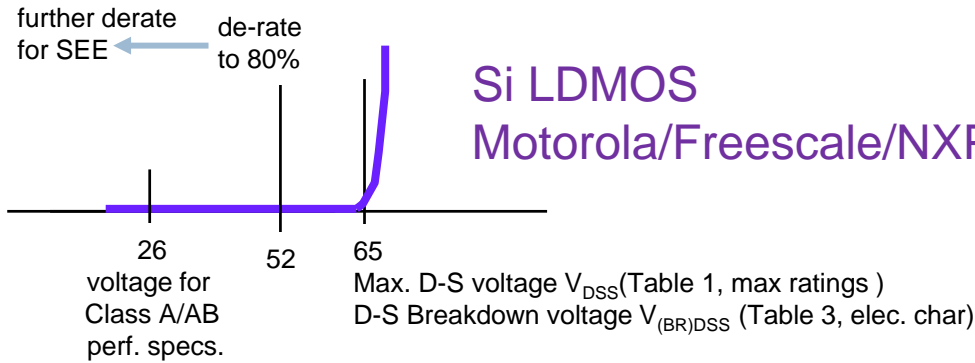
**Worsening of trapping with mission aging, stress, radiation is unknown**



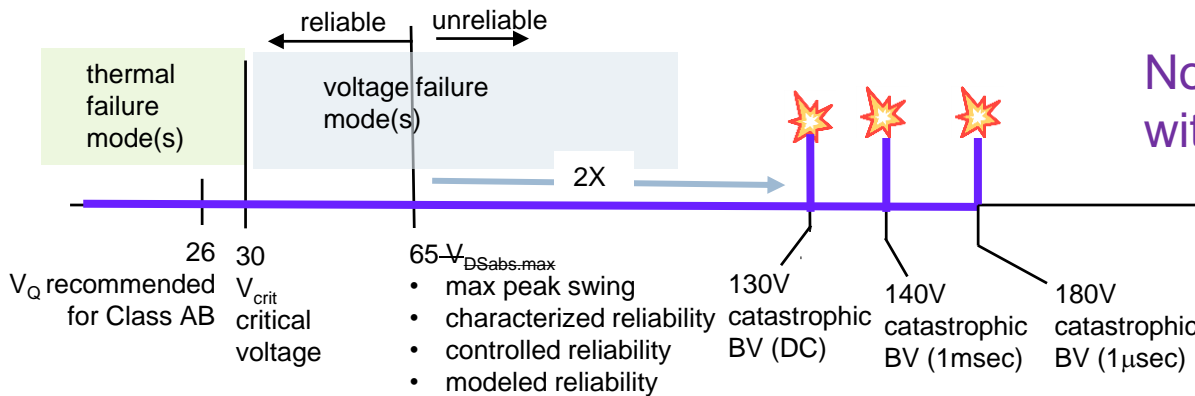
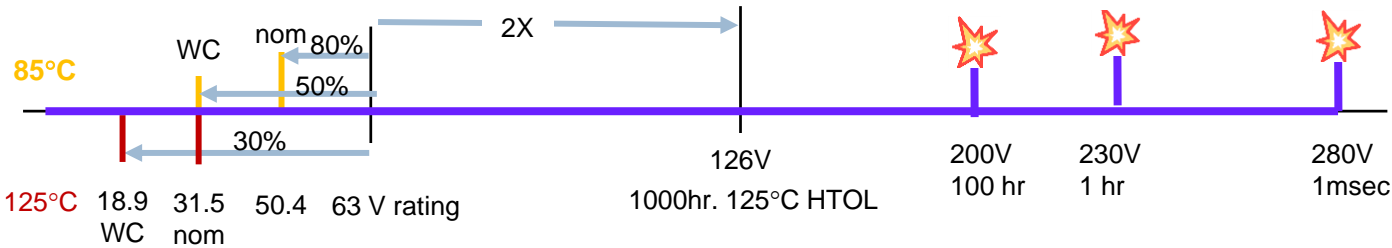


# Rating (and de-rating) Approaches

for familiar device types vs. a GaN HEMT



Multilayer Ceramic Capacitor  
0.1  $\mu$ F, 63V, 10%  
Kemet C0805C104KMRC



- GaN devices more resemble ceramic than silicon
- $V_{abs.max}$  is an unfortunate, overused and imprecise term for GaN !!

Ref.: derating rules come from TOR-2006(8583)-5236 rev. B, "Technical Requirements for Electronic Parts, Materials, and Processes Used in Space Vehicles" March 6, 2013

# Recommended RF GaN HEMT Ratings Sheet

Maximum Ratings <sup>1</sup>	Symbol	Rating	units	
Maximum channel temperature	$T_{max}$	250	°C	$T_{max}$
Maximum Safe Drain-Source Voltage <sup>2</sup>	$V_{DSmax.safe}$	70	V	$T_{max}$
Maximum Safe Drain Current per unit gate width	$I_{Dmax.safe}$	850	mA/mm	$T_{max}$
Maximum Safe Reverse Gate-Source Voltage	$V_{GRmax.safe}$	-12	V	$V_{DS} = V_{DSmax.safe} \cdot T_{max}$
Maximum Safe Gate Current per finger <sup>3</sup>	$I_{gmax.safe}$	2	mA	electromigration limit
Maximum Safe RF Input Power <sup>4</sup>	$P_{in,max.safe}$	28	dBm	CW, input return loss < 15dB
Maximum Safe RF Dissipated Power	$P_{diss,max.safe}$	33	W	CW

<sup>1</sup>Operation within these maximum ratings meets the reliability goal of 0.2% failures in 15 years with 90% confidence

<sup>2</sup>DC or pulsing above  $V_{DSmax.safe}$  reduces reliability and risks immediate or delayed catastrophic breakdown

<sup>3</sup>DC or average RF gate current

<sup>4</sup>for full recovery of performance within 1 sec after exposure

Supplementary Rating	Symbol	Rating	units	
Channel-to-Baseplate Thermal Resistance	$\theta_{CB}$	1.5	°C/W	specified at $T_{max}$
Critical Voltage <sup>5</sup>	$V_{crit}$	25	V	specified at $T_{max}$
Recommended Quiescent Voltage <sup>6</sup>	$V_Q$	35	V	
Recommended Quiescent Current <sup>6</sup>	$I_Q$	180	mA/mm	
Burnout Drain-Gate Voltage <sup>7</sup>	$V_{DG(BO)}$	140	V	1 msec DC pulse, $T = 25^\circ\text{C}$
Burnout RF Input Power <sup>7</sup>	$P_{in(BO)}$	32	dBm	1 msec RF pulse, $T = 25^\circ\text{C}$

<sup>5</sup>Operation above  $V_{crit}$  incurs performance degradations still within EOL limits if max safe ratings are obeyed.

<sup>6</sup>For typical class AB operation with compression less than 3dB

<sup>7</sup>Catastrophic burnout ensues within 1 msec of application of this applied condition



# More Recommendations & Test Protocols

*Topics to consider for space qualification of GaN*

- Robustness
  - SOA (*safe operating area*)
  - Gate burnout
  - RF burnout
  - ESD
  - Temperature cycling
  - Power cycling
  - Off-state voltage screening
- Intrinsic Reliability
  - DC lifetesting (4 Q-points)
  - RF lifetesting
  - Step stressing
  - TLYF (*Test Like You Fly*)
  - Thin film resistors
  - Electromigration
- Environmental Effects
  - Moisture sensitivity
  - Hydrogen sensitivity
  - Air Sensitivity
- Extrinsic Defects
  - MIMCAPs
  - Gate Defects
  - Airbridge Defects
  - Backside Via defects
- Mechanical
  - Backside metal adhesion
  - Bondpull tests
  - Die shear tests
  - Step Coverage
  - Low Frequency Oscillations
- Radiation Effects
  - Total Ionizing Dose
  - Dose Rate
  - Singe Event Effects SEE, SEB
  - Displacement Damage
  - GaN dosimetry, LET
- Ratings and Derating
  - Reliability, burnout and SEB considerations

***Guidelines are provided on these topics and more***



## ***Conclusion***

- A peer-reviewed and vetted space qualification methodology for GaN power HEMTs and MMICs is in press
- TOR-2018-00691 “Guidelines for Space Qualification of GaN HEMT Technologies” J. Scarpulla, C. Gee
- For more information please contact
  - *john.scarpulla@aero.org*
  - *caroline.gee@aero.org*
  
- **THANK YOU!**