Guide to Using Automotive-Grade EEEE Parts in Space Applications

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Agenda

- Introduction/Motivation
- Comparison Table (Pros and Cons)
- Comparison Matrices
 - MIL-PRF-38535 vs. AEC-Q100
 - MIL-PRF-19500 vs. AEC-Q101
 - MIL-PRF-123 vs. AEC-Q200 (Capacitors)
- Key Questions to Ask...
 - Filling Gaps Between AEC and MIL-Space Requirements
- Conclusions
- Acknowledgements

Near-Term Technology Working Group Alternate-Grade Parts for Small-Satellite Applications **Motivation** Increased use in alternate-grade • parts for space applications • Alternate-grade parts are not Alternatedesigned for space grade • AEC-qualified parts are highly Performance reliable in automotive applications (shock, vibe, thermal cycle, high temperatures) • Space-grade parts may be too expensive (upfront purchase \$) -Space-grade emphasis on lower cost launch vehicles and small satellites Technological demands – commercial parts are quicker to adapt Risk

• Space industry no longer able to dictate the EEEE parts market

Key: Risk reduction of Alternate-grade parts

Overall Comparison

Space- vs. Alternate-grade

Space-grade

Pros

High reliability in space

Little/no additional testing needed

Cons

Declining market share

Higher up-front purchase cost

Availability/long lead times

Technology/capability may lag

Alternate-grade

Pros

Greater availability

Lower up-front purchase cost

New technology/capability

Cons

Unknown reliability: may need additional testing/screening and analyses (reliability, FMEA)

Lack of data

No radiation test requirements

Trade space depends on risk posture, mission-specific environments, applicationspecific constraints, schedule, obsolescence, availability, and cost. **Some launch vehicles and small satellites currently use alternate-grade parts.**

Comparison Matrices

User Guide

- Aerospace TOR: Aiming for public release in Q3 2018
- Intended as guidelines
- Qualitative assessments by subject matter experts
- No descriptions of how parts are manufactured
- "Baseline" reference for incorporating automotive-grade parts in designs
 - Users should consider risk posture, mission-specific environments, applicationspecific constraints, schedule, obsolescence, availability, and cost
 - Users should look further into referenced MIL-PRF and MIL-STD test methods to verify whether the AEC-qual parts need additional testing.

Comparison Matrices

User Guide (Cont.)

Assessment Column:

AEC Requirement fully meets or exceeds that of the MIL-PRF
Meets intent (e.g., test might be different, but tests for same failure mechanism)
Partially meets intent
Fails to meet intent

- MIL-PRF-38535 vs. AEC-Q100: Integrated Circuits (Microcircuits)
 - Table 1A. Screening Procedure for Hermetic Classes Q, V, and Non-Hermetic Class Y
 - Table 1B. Tests/Monitors for Plastic Packages
 - Table II. Group B Tests (Mechanical and Environmental)
 - Table III. Group A (Electrical)
 - Table IV. Group C Life Tests
 - Table V. Group D Tests (Package Related)

MIL-PRF-19500 vs. AEC-Q101: Semiconductor devices

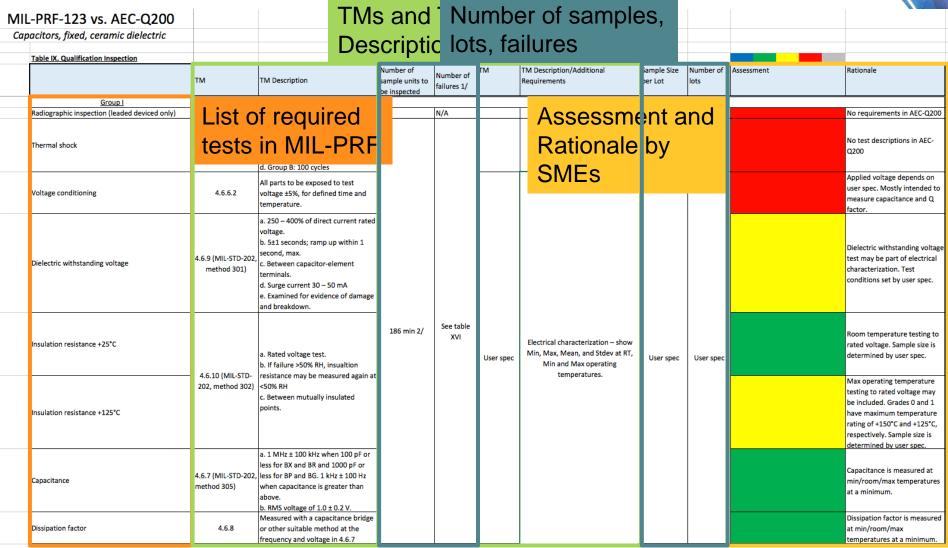
- Table E-IV. Screening Requirement
- Table E-IVA. Group B
- Table E-IVB. Group B Inspection
- Table E-V. Group A Inspection
- Table E-VII. Group C Periodic Inspection

MIL-PRF-123 vs. AEC-Q200: Capacitors, fixed, ceramic dielectric

- Table IX. Qualification Inspection

Comparison Matrices Example

Additional information found in AEC-Qxxx and PPAP documents



Organization and Features

MIL-PRF-38535 vs. AEC-Q100: ICs (Microcircuits)

Key Tests Missing in AEC-Q100



Wafer Lot Acceptance Test Quality of wafer manufacturing process



MIL-STD-883, TM 5007

- Wafer, metallization, glassivation, Au backing thicknesses
- Thermal stability
- SEM

MIL-STD-883, TM 1020

• AEC-Q100-004 Latch Up test requirements are not as stringent

Radiation Test Dose rate induced latchup test



Electrical Tests AEC does not require 100% screening Burn-in, reverse bias burn-in tests

Mostly a concern with lack of screening

MIL-PRF-38535 vs. AEC-Q100: ICs (Microcircuits)

Key Tests Missing in AEC-Q100



MIL-STD-883, TM 1005

• 1000 h minimum at 125°C

Steady-State Life Test Quality/reliability over extended time



MIL-STD-883, TM 2013

• Low- and high-magnification inspections

DPA, internal visual Internal materials, construction, workmanship



Mechanical Tests Die shear, substrate attach strength, stud pull, flip-chip pull off MIL-STD-883, TM 2014, 2004, 2038

 No AEC requirements for DPA or mechanical testing of internal bonds

MIL-PRF-19500 vs. AEC-Q101 (Semiconductors)

Key Tests Missing in AEC-Q101



MIL-STD-750, TM 1032

- ≤ Max storage T, 340 h minimum
- Optional for JANS, JANTXV, JANTX

High-Temperature Life Test

Quality/reliability over extended time



DPA, internal visual Internal materials, construction, workmanship

MIL-STD-750

- TM 2074, diodes
- TM 2069, power FETs
- TM 2070, microwave transistors
- TM 2072, transistors

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Salt Atmosphere Accelerated corrosion test

MIL-STD-750, TM 1041

+35°C salt atm for 24 +2/-0 h

MIL-PRF-19500 vs. AEC-Q101 (Semiconductors)

Key Tests Missing in AEC-Q101



Moisture Resistance Resistance to high humidity and heat



MIL-STD-750, TM 1018

MIL-STD-750, TM 1021

Gases inside hermetically-sealed packages can affect reliability

Internal Gas Analysis Gas atmosphere inside hermetic devices

MIL-PRF-123 vs. AEC-Q200 (Capacitors)

Key Tests Missing in AEC-Q200



MIL-PRF-123, sec 4.6.5

All leaded devices are inspected

Radiographic Inspection Inspection for defects



MIL-STD-202, TM 107

- Tested at +125°C
- AEC-Q200 does not include test conditions

Thermal Shock Resistance to temperature extremes



Help eliminate infant mortality

MIL-PRF-123, sec 4.6.6.2

- All parts to be exposed to test voltage ±5% for defined time and temperature
- AEC-Q200: depends on user spec

MIL-PRF-123 vs. AEC-Q200 (Capacitors)

Key Tests Missing in AEC-Q200



Terminal Strength Determine integrity of terminals



Life Test Quality/reliability over extended time

MIL-PRF-123, sec 4.6.5

- All leaded devices are inspected
- AEC-Q200 does not require testing of non-leaded devices

MIL-STD-202, TM 108

- Tested at +125°C, 4000 h (qual) and 1000 h (Group B)
- AEC-Q200 does not include test conditions

Key Questions to Ask...

Filling Gaps Between AEC and MIL-Space Requirements

Lack of 100% screening

- Demonstrate lot homogeneity and device consistency (e.g., Cpk, Ppk)
 - Cpk alone does not necessarily demonstrate this both Cpk and Ppk would be ideal, but may not be always available.
- Parts currently available vs. future builds
- Qualification by similarity
- Screening test data
- Verification schedules for AEC qualification
- Origin of design, manufacturing, packaging, and testing
- Perform DPA (third party vendor or in-house)

Tailoring to mission conditions

- Radiation environment
- Mission duration
- Potential single-point failure and mission criticality

Key Questions to Ask...

Filling Gaps Between AEC and MIL-Space Requirements

- Insufficient testing (qual or screen) requirements
 - Operating temperature range of device
 - Stress test results and failure mechanism information
 - Supplier's reliability test criteria
 - FIT (failures in time) data
 - Test to failure data
 - Identification and elimination of potential causes of defects
 - FMEA, corrective and preventive actions
 - Flight heritage information
 - Supplier audits (if possible)
 - Level of process and materials changes that trigger advanced notifications
 - Priority problem resolution or end-of-life notifications
 - Lessons learned or early alert system
 - Prevention of counterfeit parts (when purchased through third-party vendors)
 - Maintain preferred vendors list

Conclusions

Holistic Approach to Parts Selection

- High reliability is no accident!
 - Collect as much information/data available from the supplier
 - Assess process repeatability
 - Maintain preferred suppliers list
 - Perform additional testing as needed
 - Radiation
 - DPA
- AEC-Qxxx: Qualification requirements, not screening screening may be performed as part of manufacturer's own requirements or as an agreement between the manufacturer and its customer (PPAP).
 - Level of testing/assurance may vary amongst part lots.
- Tailoring to mission conditions
 - E.g., Short missions may not require extensive radiation or lifetime testing
 - Redundancies costs of launch services are decreasing
 - Single, high-cost mission vs. multiple, lower-cost vehicles in constellation
 - Technological needs do space-grade parts provide sufficient capabilities?

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